Energy-Based Control of Multicopters

Task and motivation

Multicopters

- UAVs with multiple propellers
- increasingly popular in hobby and industry

State of the art

- position controlled by a pilot
- *linear* stabilization of attitude
- \hookrightarrow not capable of large tilt angles

Goal

 nonlinear tracking control of position and attitude capable of aerobatic maneuvers

Experimental setup







Lagrange's formalism

Redundant configuration coordinates x(t) and configuration manifold $\mathbb X$

$$\mathbb{X} = \{ x \in \mathbb{R}^{\nu} \, | \, \phi(x) = 0 \}, \qquad n = \dim \mathbb{X} = \nu - \operatorname{rank} \frac{\partial \phi}{\partial x}$$

Minimal velocity coordinates $\xi(t) \in \mathbb{R}^n$ and kinematics

$$\dot{x}^{\alpha} = A_i^{\alpha}(x)\xi^i, \ \alpha = 1, \dots, \nu$$
 $\frac{\partial \phi^{\kappa}}{\partial x^{\alpha}}A_i^{\alpha} \equiv 0, \ \operatorname{rank} A = n$

Directional derivative ∂_i and commutator coefficients γ_{ij}^k

$$\partial_i = A_i^{\alpha} \frac{\partial}{\partial x^{\alpha}}, \qquad \partial_i \partial_j - \partial_j \partial_i = \gamma_{ij}^k \partial_k, \quad \gamma_{ij}^k = (\partial_i A_j^{\alpha} - \partial_j A_i^{\alpha}) (A^+)_{\alpha}^i$$

Lagrange's equation

$$\frac{\mathrm{d}}{\mathrm{d}t}\frac{\partial L}{\partial\xi^{i}} + \gamma_{ij}^{k}\xi^{j}\frac{\partial L}{\partial\xi^{k}} - \partial_{i}L = f_{i}, \ i = 1, \dots, n$$

[M. Konz, J. Rudolph: *Equations of Motion with Redundant Coordinates for Mechanical Systems on Manifolds*, Proc. 8th Vienna International Conference on Mathematical Modelling, pp. 697-698, 2015]

Rigid body model

Coordinates

$$x \cong (r, R) \in \mathbb{R}^3 \times \mathsf{SO}(3), \ \phi(x) \cong R^\top R - I_3 = 0$$

$$\xi \cong (v, \omega) \in \mathbb{R}^6$$

Kinematics

$$\dot{r} = Rv, \ \dot{R} = R\widehat{\omega} \quad \cong \quad \dot{x} = A(x)\xi$$

Kinetics from Lagrange's equation



Microcontroller

- \blacktriangleright ATMEL AVR32UC3C: 32bit, 66MHz, FPU, compiler for C and C++
- sampling time for controller implementation 5 ms

Some practical challenges

- ► Inertial meas. unit (IMU): gyroscope and accelerometer
- \blacktriangleright VICON motion capture sys. (Mo-Cap): accurate but delayed \approx 70 ms
- \hookrightarrow dynamic, model-based fusion for real-time state estimate
- Propellers are driven by brushless DC motors (BLDC)
- $\hookrightarrow \mathsf{Fast}$ underlying control of propeller speed

[D. Kastelan, M. Konz, J. Rudolph: *Fully actuated tricopter with pilot-supporting control*, IFAC-PapersOnLine 28(9), pp. 79–84, 2015]

Theoretical challenges

Rigid body dynamics are the dominant part of the system

- \blacktriangleright unstable system \Rightarrow feedback control is inevitable
- \blacktriangleright nonlinear configuration manifold \Rightarrow minimal parameterizations are only local

Experimental result

Tracking a circular reference trajectory (velocity on circle pprox 5 m/s, max. acceleration pprox 18 m/s²)

$M(x)\dot{\xi} + c(x,\xi) + \nabla V(x) = Bu$ Desired controlled system

- \blacktriangleright desired inertia $\bar{M}(x)>0$
- \blacktriangleright desired dissipation $\bar{D}(x) \geq 0$
- \blacktriangleright desired potential $\bar{V}(x) \geq 0$, $\bar{V}=0 \ \Leftrightarrow \ x=x_{\mathsf{R}}$
- \hookrightarrow closed loop kinetics

$$\bar{M}(x)\dot{\xi} + \bar{c}(x,\xi) + \bar{D}(x)\xi + \nabla\bar{V}(x) = 0$$

Total energy $ar{H}$ is a Lyapunov function

$$\bar{H} = \frac{1}{2}\xi^{\top}\bar{M}\xi + V \ge 0, \quad \dot{\bar{H}} = -\xi^{\top}\bar{D}\xi \le 0$$

Control law

Insert desired kinetics into the model kinetics to get required generalized force

$$f = c + D\xi + \nabla V - M\bar{M}^{-1}(\bar{c} + \bar{D}\xi + \nabla\bar{V})$$

 \hookrightarrow control law and matching condition

$$\iota = B^+ f, \qquad B^\perp f \stackrel{!}{=} 0$$

Extension to *tracking control* is more sophisticated but follows the same basic idea [M. Konz, J. Rudolph: *Beispiele für einen direkten Zugang zu einer globalen, energiebasierten Modellbildung und Regelung von Starrkörpersystemen,* at - Automatisierungstechnik 64(2), pp. 96–109, 2016]











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